## We claim:

- 1 1. A thruster for a vehicle comprising:
- 2 a source of propellant;
- a source of microwave energy at or above approximately 35 GHz;
- 4 a heat exchanger for receiving energy from the source of microwave energy and
- 5 coupled to the source of propellant to heat the propellant to create an expanding
- 6 medium; and
- 7 a thrust converter coupled to the heat exchanger to generate thrust from the
- 8 expanding medium.
- 1 2. The thruster of claim 1 where the source of propellant comprises a source of
- 2 hydrogen gas under pressure.
- 1 3. The thruster of claim 1 where the source of propellant comprises ambient
- 2 atmosphere.
- 1 4. The thruster of claim 1 where source of microwave energy is at or above
- 2 approximately 140 GHz.

- 1 5. The thruster of claim 1 where the source of propellant comprises ambient
- 2 atmosphere and a combustible material, which is combusted with the ambient
- 3 atmosphere to release additional energy added to that from the source of microwave
- 4 energy and delivered to the thrust converter.
- 1 6. The thruster of claim 1 where the heat exchanger comprises a microwave
- 2 absorber thermally coupled with the propellant supplied from the source of propellant
- and electromagnetically coupled to the source of microwave energy.
- 1 7. The thruster of claim 4 where the microwave absorber comprises a lossy
- 2 dielectric block having a plurality of propellant channels defined therethrough.
- 1 8. The thruster of claim 4 where the microwave absorber comprises a susceptor.
- 1 9. The thruster of claim 1 where the susceptor comprises a sheet having a layered
- 2 structure in which a plurality of propellant channels are defined and in which a susceptor
- 3 is embedded.
- 1 10. The thruster of claim 9 where the layered structure comprises a susceptor layer
- 2 disposed about the plurality of propellant channels and in intimate heat exchanging
- 3 relationship with propellant flowing through the plurality of propellant channels, a
- 4 dielectric layer disposed on each side of the susceptor layer, and a reflector layer

- 5 adjacent the dielectric layer on the side opposite the dielectric layer providing a
- 6 microwave exposure surface, the dielectric, susceptor and reflector layers chosen in
- 7 material properties and dimensions to maximize power absorption fraction of the
- 8 susceptor layer.
- 1 11. A method of propelling a vehicle comprising:
- 2 providing propellant in the vehicle;
- 3 providing microwave energy broadcast to the vehicle at or above 35 GHz;
- 4 absorbing the microwave energy in a body of a microwave absorbing heat
- 5 exchanger in the vehicle;
- 6 transferring the absorbed energy in the body of the heat exchanger to the
- 7 propellant; and
- 8 converting the energized propellant into thrust for the vehicle.
- 1 12. The method of claim 11 where providing propellant comprises providing
- 2 hydrogen.
- 1 13. The method of claim 11 where providing propellant comprises providing ambient
- 2 atmosphere.
- 1 14. The method of claim 12 where providing microwave energy provides the
- 2 microwave energy at or above 140 GHz.

- 1 15. The method of claim 11 where providing propellant comprises providing ambient
- 2 atmosphere and a combustible material, combusting the combustible material with the
- 3 ambient atmosphere to release additional energy added to that from the source of
- 4 microwave energy and converting the additional energy to thrust.
- 1 16. The method of claim 11 where absorbing the microwave energy comprises
- 2 electromagnetically coupling a heat exchanger with a source of microwave energy and
- 3 where transferring the absorbed energy to the propellant comprises transferring the
- 4 absorbed microwave energy to the propellant supplied by means of a flow heat
- 5 exchanger.
- 1 17. The method of claim 14 where absorbing the microwave energy comprises
- 2 absorbing the microwave energy in a lossy dielectric structure.
- 1 18. The method of claim 14 where absorbing the microwave energy comprises
- 2 absorbing the microwave energy in a susceptor.
- 1 19. The method of claim 11 where absorbing the microwave energy comprises
- 2 absorbing the energy by means of a sheet having a layered structure in which a plurality
- 3 of propellant channels are defined and in which a susceptor is embedded.

- 1 20. The method of claim 11 where absorbing the energy by means of a sheet having
- 2 a layered structure comprises exposing a first dielectric layer to microwave energy,
- 3 absorbing the energy by means of a first and second susceptor layer disposed beneath
- 4 the first dielectric layer, exchanging heat with propellant flowing through the plurality of
- 5 propellant channels in intimate contact with the first and second susceptor layers,
- 6 supporting the second susceptor layer with a second a dielectric layer, backing the
- 7 second susceptor layer with a reflector layer, and choosing the dielectric, susceptor and
- 8 reflector layers in terms of both material properties and dimensions to maximize power
- 9 absorption fraction of the susceptor layer.
- 10 21. The method of claim 11 where a microwave thruster forms an undersurface of the vehicle, and where providing and absorbing the microwave energy comprises intercepting microwaves from a ground-based source or plurality of sources with the undersurface of the craft such that sufficient energy necessary to achieve a given trajectory is received at every point along the trajectory.
- 15 22. The thruster of claim 1 further comprising an aeroshell for the vehicle and where the thruster forms part of the aeroshell.
  - 23. The thruster of claim 22 where the microwave thruster acts as part of an atmospheric re-entry heat shield.

- 24. The thruster of claim 23 further comprising a source of cooling fluid coupled to the thruster and where a fluid is selectively passed through the heat exchanger to cool the overall structure.
- 25. The thruster of claim 8 where the susceptor is comprised of at least one layerhaving an electroplated surface.
  - 26. The thruster of claim 25 where the susceptor is comprised of a plurality of susceptor layers, which are clad with one or more dielectric and reflecting layers, and where at least one of the susceptor, dielectric and reflecting layers are vapor deposited.
- The method of claim 11 where providing the microwave energy comprises
  providing microwave energy at a source frequency, and further comprises tuning the source frequency to match maximum absorption to help the absorbing exchanger to heat up.
  - 28. The method of claim 27 where tuning the source frequency comprises continuously sweeping a range of frequencies to promote uniformity of heating when the absorbing heat exchanger is operating in or close to a thermal steady state.
  - 29. The thruster of claim 7 where the thrust converter comprises a Laval nozzle.

15

- 30. The thruster of claim 8 where the thrust converter comprises a Plug nozzle.
- 31. The thruster of claim 1 where the source of propellant is a source of hydrogen.
- 32. The thruster of claim 1 where the source of propellant is a source of ammonia.
- 33. The method of claim 11 further comprising heating dummy warheads after separation from a delivery vehicle by means of the broadcast microwave energy and sensing the dummy warheads by infrared signatures thereof created by the differential heating between dummy and real warheads.